



14. Mechanical properties of Materials



Different mechanical tests

- Tension
- Compression
- Shearing

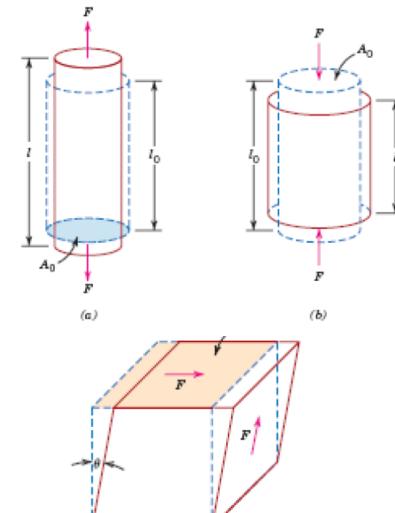




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DEPARTMENT OF MECHANICAL AND AUTOMATION ENGINEERING

Stress vs. Strain



(c)

Normal stress

$$\sigma = \frac{F}{A_0}$$

Normal strain

$$\varepsilon = \frac{l - l_0}{l_0} \times 100\%$$

Shear stress

$$\tau = \frac{F}{A_0}$$

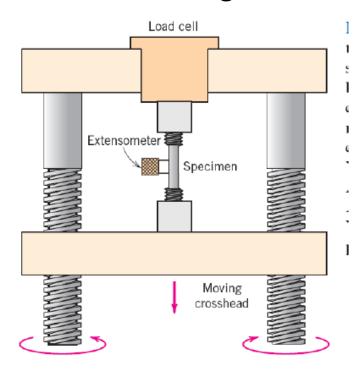
Shear strain

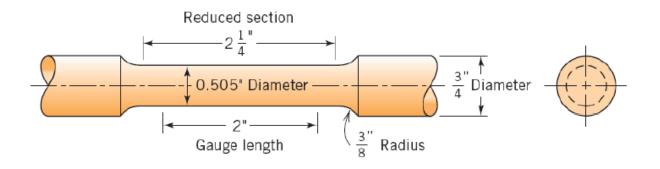
$$\varepsilon = \tan \theta$$





Stress-Strain diagram

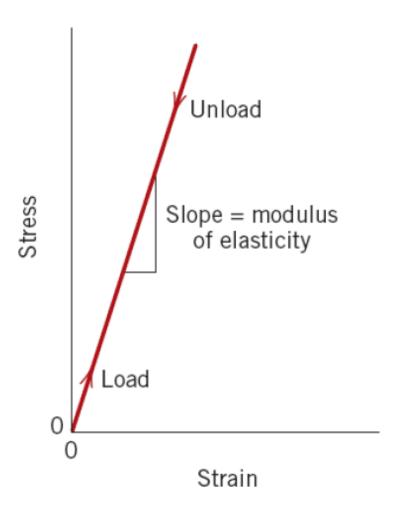


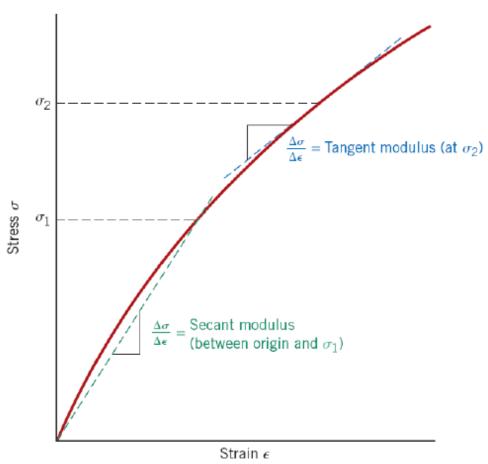






Elasticity







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	Modulus of Elasticity		Shear Modulus		
Material	GPa	10 ⁶ psi	GPa	106 psi	Poisson's Rati
	Meta	l Alloys			
Tungsten	407	59	160	23.2	0.28
Steel	207	30	83	12.0	0.30
Nickel	207	30	76	11.0	0.31
Titanium	107	15.5	45	6.5	0.34
Copper	110	16	46	6.7	0.34
Brass	97	14	37	5.4	0.34
Aluminum	69	10	25	3.6	0.33
Magnesium	45	6.5	17	2.5	0.35
	Ceramic	Materials			
Aluminum oxide (Al ₂ O ₃)	393	57	1-	-	0.22
Silicon carbide (SiC)	345	50	-	-	0.17
Silicon nitride (Si ₃ N ₄)	304	44			0.30
Spinel (MgAl ₂ O ₄)	260	38	_	-	-
Magnesium oxide (MgO)	225	33	_	_	0.18
Zirconia (ZrO ₂) ^a	205	30	-	_	0.31
Mullite (3Al ₂ O ₃ -2SiO ₂)	145	21	_	_	0.24
Glass-ceramic (Pyroceram)	120	17	-	_	0.25
Fused silica (SiO ₂)	73	11	-		0.17
Soda-lime glass	69	10			0.23
	Poly	ymers ^b			
Phenol-formaldehyde	2.76-4.83	0.40-0.70	-	_	1-1
Poly(vinyl chloride) (PVC)	2.41-4.14	0.35-0.60	-	_	0.38
Poly(ethylene terephthalate) (PET)	2.76-4.14	0.40-0.60	-	_	0.33
Polystyrene (PS)	2.28-3.28	0.33-0.48	1		0.33
Poly(methyl methacrylate) (PMMA)	2.24-3.24	0.33-0.47	1		0.37-0.44
Polycarbonate (PC)	2.38	0.35	-	-	0.36
Nylon 6,6	1.59-3.79	0.23-0.55	-	-	0.39
Polypropylene (PP)	1.14-1.55	0.17-0.23	-	-	0.40
Polyethylene-high density (HDPE)	1.08	0.16	-	-	0.46
Polytetrafluoroethylene (PTFE)	0.40-0.55	0.058-0.080	-		0.46
Polyethylene-low density (LDPE)	0.17-0.28	0.025-0.041	-	-	0.33-0.40

Partially stabilized with 3 mol% Y2O3.

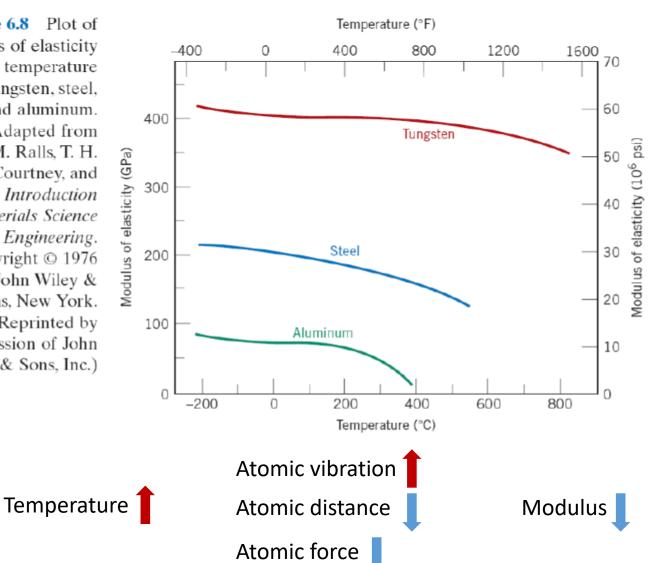
^bModern Plastics Encyclopedia '96, McGraw-Hill, New York, 1995.





Temperature-dependent modulus

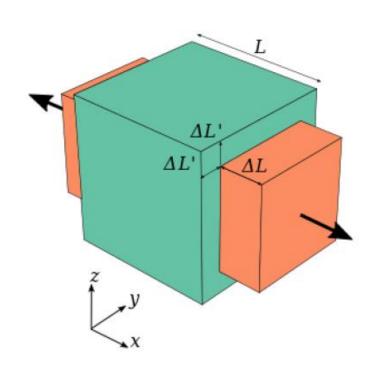
Figure 6.8 Plot of modulus of elasticity versus temperature for tungsten, steel. and aluminum. (Adapted from K. M. Ralls, T. H. Courtney, and J. Wulff, Introduction to Materials Science and Engineering. Copyright © 1976 by John Wiley & Sons, New York. Reprinted by permission of John Wiley & Sons, Inc.)







Poisson's Ratio



$$darepsilon_x = rac{dx}{x} \qquad darepsilon_y = rac{dy}{y} \qquad darepsilon_z = rac{dz}{z}$$

$$-
u\int_{L}^{L+\Delta L}rac{dx}{x}=\int_{L}^{L+\Delta L'}rac{dy}{y}=\int_{L}^{L+\Delta L'}rac{dz}{z}$$

$$\left(1+rac{\Delta L}{L}
ight)^{-
u}=1+rac{\Delta L'}{L}$$

$$u pprox -rac{\Delta L'}{\Delta L}$$





Inelasticity

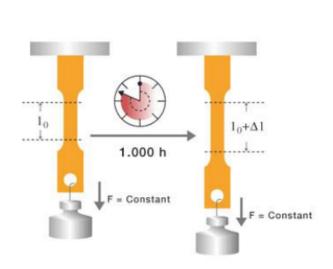
Elastic deformation can be time-dependent:

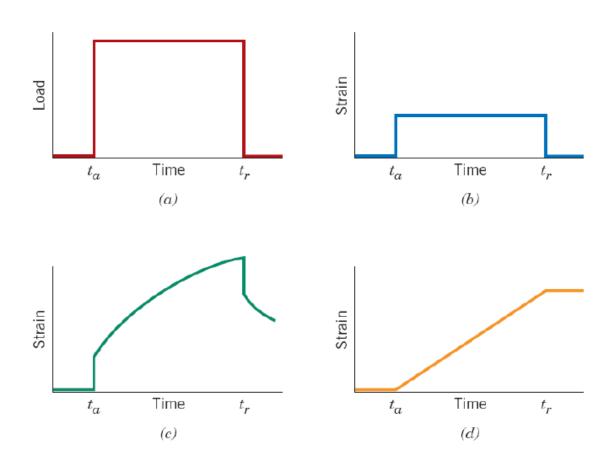
- With constant stress, the strain may further increase;
- After stress is removed, strain goes back to zero after a while;
- For polymers, such time-dependent behavior is significant: Viscoelasticity





Viscoelasticity: Creep

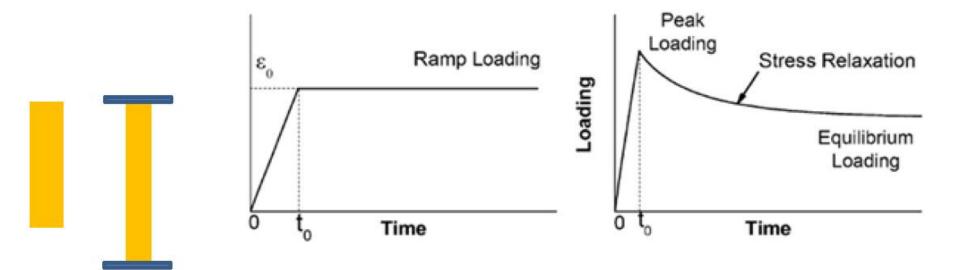








Viscoelasticity: Relaxation



Fix deformation

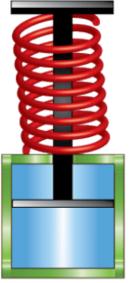




Spring dashpot for viscoelasticity

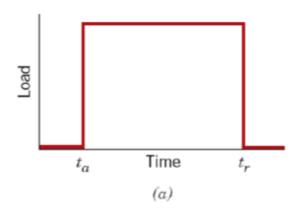
Conceptual model

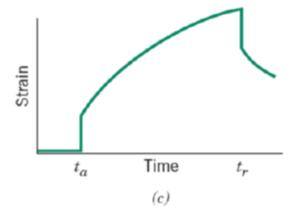
Viscoelastic Model

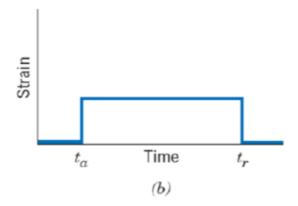


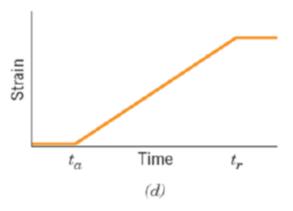
Spring models elasticity

Dashpot models viscosity





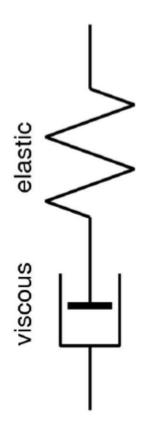








Viscoelasticity Model



$$egin{aligned} \sigma_{ ext{Total}} &= \sigma_D = \sigma_S \ arepsilon_{ ext{Total}} &= arepsilon_D + arepsilon_S \ \ rac{darepsilon_{ ext{Total}}}{dt} &= rac{darepsilon_D}{dt} + rac{darepsilon_S}{dt} = rac{\sigma}{\eta} + rac{1}{E}rac{d\sigma}{dt} \end{aligned}$$

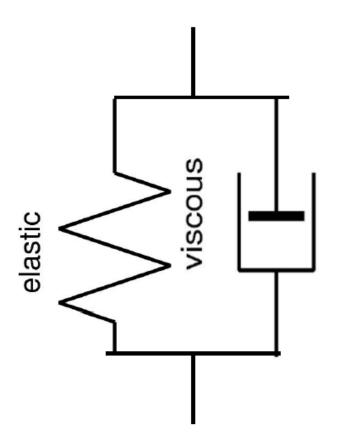
$$rac{\dot{\sigma}}{E}+rac{\sigma}{\eta}=\dot{arepsilon}$$

Maxwell model





Viscoelasticity Model



$$\varepsilon_{\mathrm{Total}} = \varepsilon_S = \varepsilon_D$$

$$\sigma_{\mathrm{Total}} = \sigma_S + \sigma_D$$

$$\sigma(t) = E arepsilon(t) + \eta rac{darepsilon(t)}{dt}$$

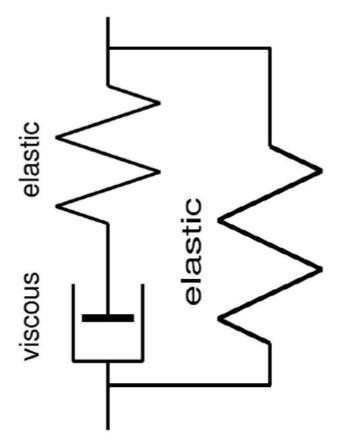
$$\sigma = E \varepsilon + \eta \dot{arepsilon}$$

Kelvin-Voigt model





Viscoelasticity Model



$$egin{aligned} \sigma_{tot} &= \sigma_m + \sigma_{S_1} \ arepsilon_{tot} &= arepsilon_m = arepsilon_{S_1} \ \sigma_m &= \sigma_D = \sigma_{S_2} \ arepsilon_m &= arepsilon_D + arepsilon_{S_2} \ arepsilon_m &= arepsilon_D + arepsilon_{S_2} \ arepsilon_m &= \dfrac{rac{E_2}{\eta} \left(rac{\eta}{E_2} rac{d\sigma(t)}{dt} + \sigma(t) - E_1 arepsilon(t)
ight)}{E_1 + E_2} \end{aligned}$$

$$\sigma(t)+rac{\eta}{E_2}rac{d\sigma(t)}{dt}=E_1arepsilon(t)+rac{\eta(E_1+E_2)}{E_2}rac{darepsilon(t)}{dt}$$

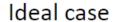
$$\sigma+rac{\eta}{E_1+E_2}\dot{\sigma}=rac{E_1E_2}{E_1+E_2}arepsilon+rac{E_1\eta}{E_1+E_2}\dot{arepsilon}$$

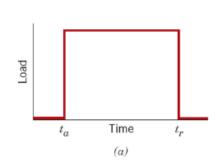
Zener model

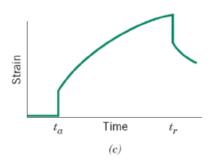


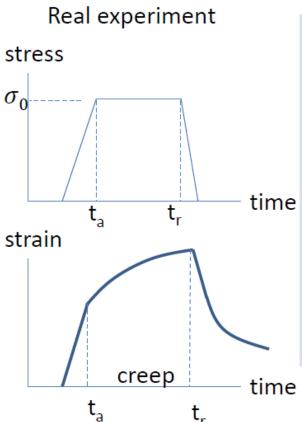


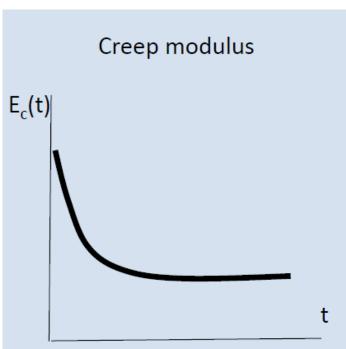
Creep experiment











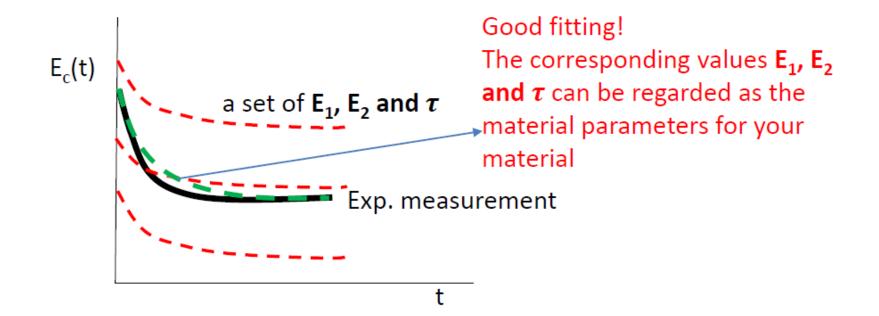




Data fittings

Zener model:
$$E_c(t) = \frac{1}{\frac{1}{E_1 + E_2} + \left(\frac{1}{E_1} - \frac{1}{E_1 + E_2}\right) \left(1 - \exp\left(-\frac{t}{\tau}\right)\right)}$$

Selecting a group of material parameters: E_1 , E_2 and τ , to make the prediction of the model close to experimental measurements



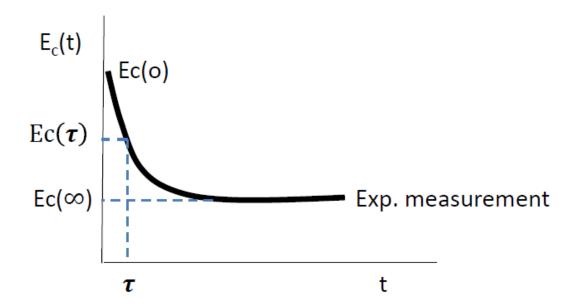




Rough estimations of material parameters in the model

$$E_c(t) = \frac{1}{\frac{1}{E_1 + E_2} + \left(\frac{1}{E_1} - \frac{1}{E_1 + E_2}\right) \left(1 - \exp\left(-\frac{t}{\tau}\right)\right)},$$

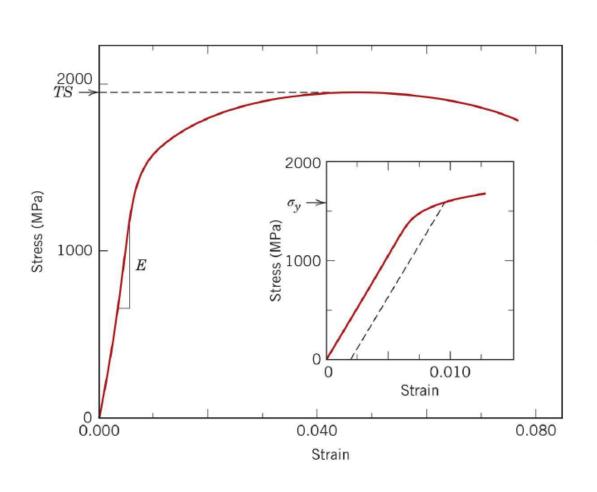
When t=0, Ec(o)= $\mathbf{E_1}$ + $\mathbf{E_2}$; when t= ∞ , Ec(∞)= $\mathbf{E_1}$; When t= τ , Ec(τ) = $(\mathbf{E_1}$ + $\mathbf{E_2})/(1+(1-e^{-1})(\mathbf{E_2}/\mathbf{E_1}))$;

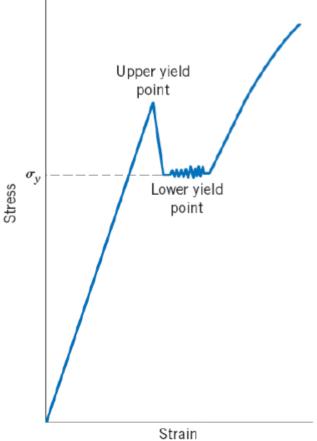






Plastic deformation

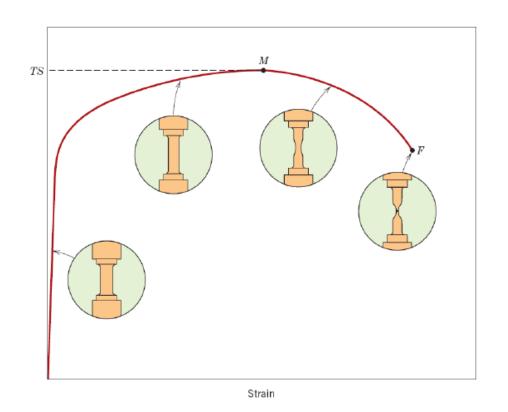


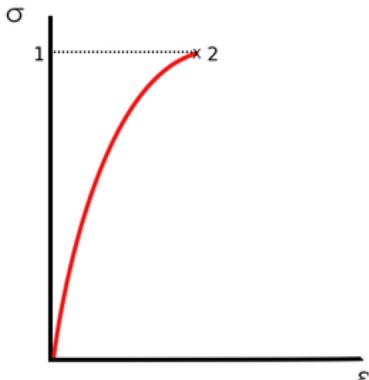






Strength



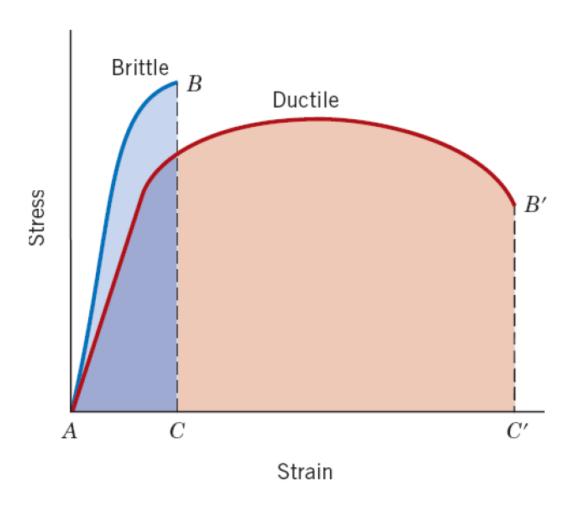


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Ductility, Resilience and Toughness







Ductility, Resilience and Toughness

	Yield Strength		Tensile Strength		Ductility, %EI
Material	MPa	ksi	MPa	ksi	$[in 50 mm (2 in.)]^a$
	Metal	Alloys ^b			
Molybdenum	565	82	655	95	35
Titanium	450	65	520	75	25
Steel (1020)	180	26	380	55	25
Nickel	138	20	480	70	40
Iron	130	19	262	38	45
Brass (70 Cu-30 Zn)	75	11	300	44	68
Copper	69	10	200	29	45
Aluminum	35	5	90	13	40
	Ceramic	Materialsc			
Zirconia (ZrO ₂) ^d		1-0	800-1500	115-215	
Silicon nitride (Si ₃ N ₄)	-	-	250-1000	35-145	_
Aluminum oxide (Al ₂ O ₃)	-		275-700	40-100	_
Silicon carbide (SiC)	_	_	100-820	15-120	_
Glass-ceramic (Pyroceram)	-	(-0)	247	36	_
Mullite (3Al ₂ O ₃ -2SiO ₂)	-	9_8	185	27	
Spinel (MgAl ₂ O ₄)	22	-	110-245	16-36	
Fused silica (SiO ₂)	_	_	110	16	-
Magnesium oxide (MgO) ^e	-	_	105	15	_
Soda-lime glass	-	1-1	69	10	-
	Poly	mers			
Nylon 6,6	44.8-82.8	6.5-12	75.9-94.5	11.0-13.7	15-300
Polycarbonate (PC)	62.1	9.0	62.8-72.4	9.1-10.5	110-150
Poly(ethylene terephthalate) (PET)	59.3	8.6	48.3-72.4	7.0-10.5	30-300
Poly(methyl methacrylate) (PMMA)	53.8-73.1	7.8-10.6	48.3-72.4	7.0-10.5	2.0-5.5
Poly(vinyl chloride) (PVC)	40.7-44.8	5.9-6.5	40.7-51.7	5.9-7.5	40-80
Phenol-formaldehyde	_	-	34.5-62.1	5.0-9.0	1.5-2.0
Polystyrene (PS)	25.0-69.0	3.63-10.0	35.9-51.7	5.2-7.5	1.2-2.5
Polypropylene (PP)	31.0-37.2	4.5-5.4	31.0-41.4	4.5-6.0	100-600
Polyethylene - high density (HDPE)	26.2-33.1	3.8-4.8	22.1-31.0	3.2-4.5	10-1200
Polytetrafluoroethylene (PTFE)	13.8-15.2	2.0-2.2	20.7-34.5	3.0-5.0	200-400
Polyethylene - low density (LDPE)	9.0-14.5	1.3-2.1	8.3-31.4	1.2-4.55	100-650

[&]quot;For polymers, percent elongation at break.

^bProperty values are for metal alloys in an annealed state.

^{&#}x27;The tensile strength of ceramic materials is taken as flexural strength (Section 7.10).

^dPartially stabilized with 3 mol% Y₂O₃.

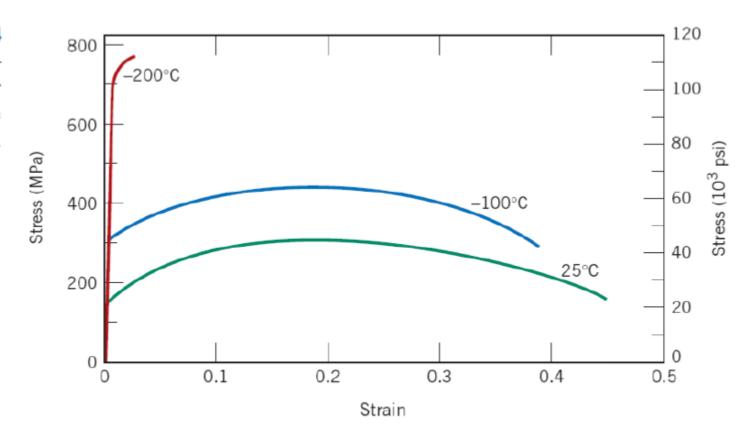
[&]quot;Sintered and containing approximately 5% porosity.





Temperature-dependent mechanical properties

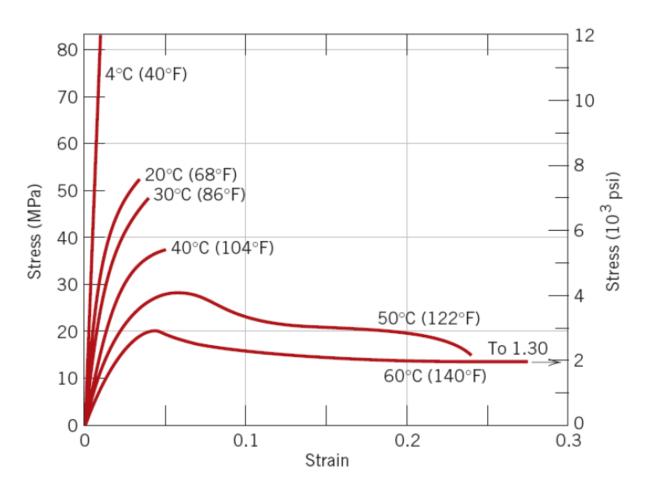
Figure 6.14
Engineering stress–
strain behavior
for iron at three
temperatures.







Temperature-dependent mechanical properties

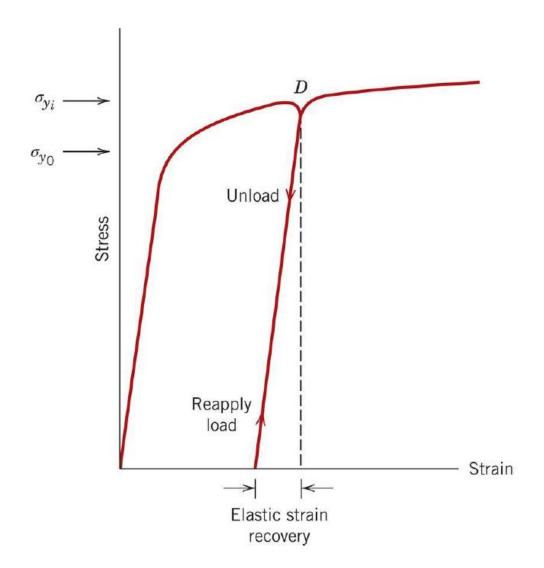


PMMA





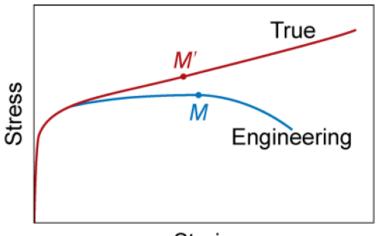
Elastic recovery after plastic deformation







True stress and true strain



$$\sigma_T = K(\varepsilon_T)^n$$

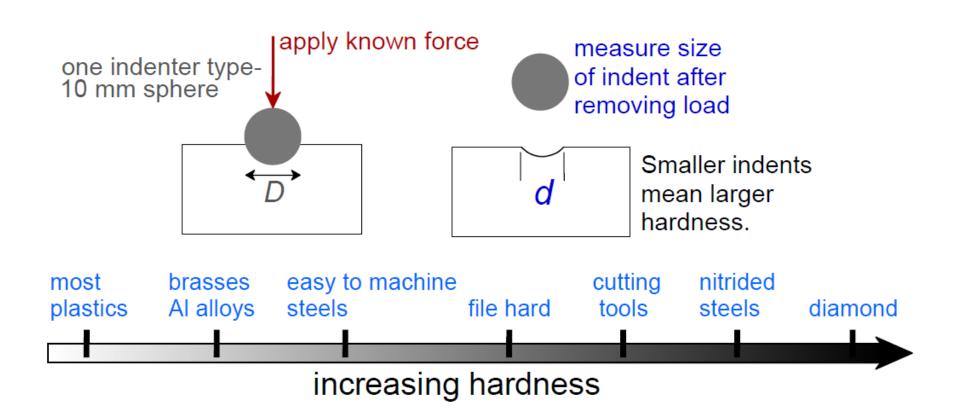
Strain

		K		
Material	n	MPa	psi	
Low-carbon steel (annealed)	0.21	600	87,000	
4340 steel alloy (tempered at 315°C)	0.12	2650	385,000	
304 stainless steel (annealed)	0.44	1400	205,000	
Copper (annealed)	0.44	530	76,500	
Naval brass (annealed)	0.21	585	85,000	
2024 aluminum alloy (heat-treated—T3)	0.17	780	113,000	
AZ-31B magnesium alloy (annealed)	0.16	450	66,000	





Hardness







Hardness

Test	Indenter	Shape of Indentation			Formula for
		Side View	Top View	Load	Hardness Number ^a
Brinell	10-mm sphere of steel or tungsten carbide			P	$HB = \frac{2P}{\pi D \left[D - \sqrt{D^2 - d^2}\right]}$
Vickers microhardness	Diamond pyramid	136°	d_1 d_1	P	$HV = 1.854 P/d_1^2$
Knoop microhardness	Diamond pyramid _	<i>l/b</i> = 7.11 <i>b/t</i> = 4.00	b	P	$HK = 14.2 P/l^2$
Rockwell and superficial Rockwell	$\begin{cases} \text{Diamond cone;} \\ \frac{1}{16}\text{-}, \frac{1}{8}\text{-}, \frac{1}{4}\text{-}, \frac{1}{2}\text{-} \text{ in.} \\ \text{diameter} \\ \text{tungsten carbide} \\ \text{spheres} \end{cases}$	120°		$ \begin{pmatrix} 60 \text{ kg} \\ 100 \text{ kg} \\ 150 \text{ kg} \end{pmatrix} $ Rockwell $ \begin{pmatrix} 15 \text{ kg} \\ 30 \text{ kg} \\ 45 \text{ kg} \end{pmatrix} $ Superficial Rockwell	

[&]quot;For the hardness formulas given, P (the applied load) is in kg and D, d, d_1 , and l are all in mm.

Source: Adapted from H. W. Hayden, W. G. Moffatt, and J. Wulff, *The Structure and Properties of Materials*, Vol. III, *Mechanical Behavior*. Copyright © 1965 by John Wiley & Sons, New York. Reprinted by permission of John Wiley & Sons, Inc.



Variability of Material Properties

- Statistical treatments
- Typical value—take average value, X for some parameter x:

$$n = \text{number of measurements}$$
 $\overline{X} = \frac{\sum_{i=1}^{n} X_i}{n}$ $X_i = \text{specific measured value}$

• Degree of scatter—use standard deviation, s

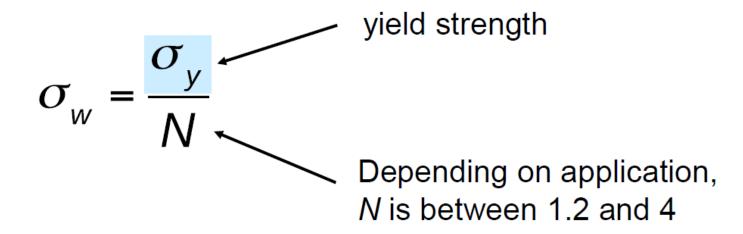
$$S = \left[\frac{\sum_{i=1}^{n} (x_i - \overline{x})^2}{n-1}\right]^{\frac{1}{2}}$$





Design/Safety Factors

- Because of design uncertainties allowances must be made to protect against unanticipated failure
- For structural applications, to protect against possibility of failure—use working stress, σ_w, and a factor of safety, N

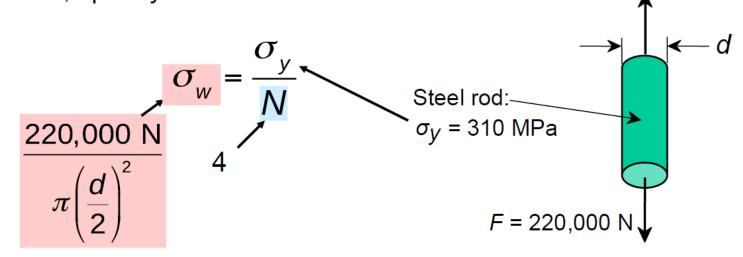






Design/Safety Factors

Example Problem: A cylindrical rod, to be constructed from a steel that has a yield strength of 310 MPa, is to withstand a load of 220,000 N without yielding. Assuming a value of 4 for N, specify a suitable bar diameter.



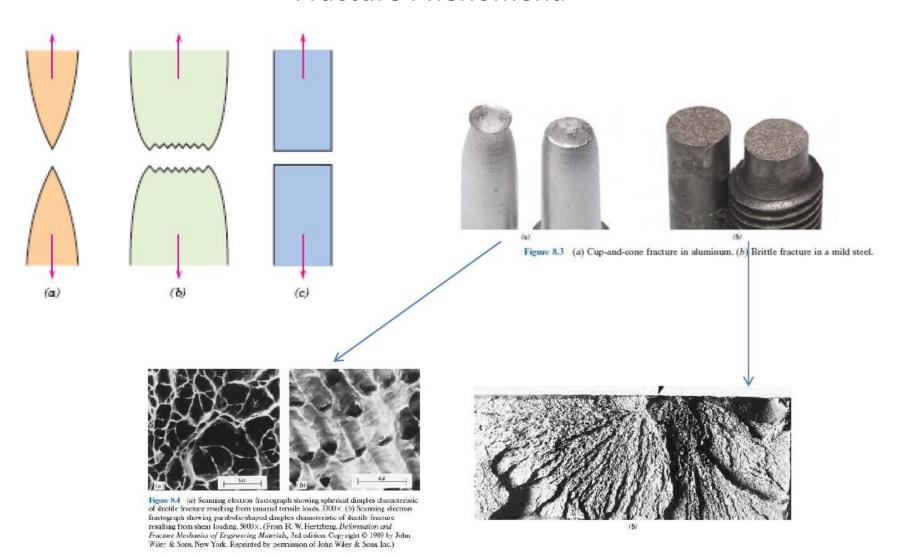
Solving for the rod diameter *d* yields

$$d = 0.060 \text{ m} = 60 \text{ mm}$$





Fracture Phenomena







Microscopic process of fracturing

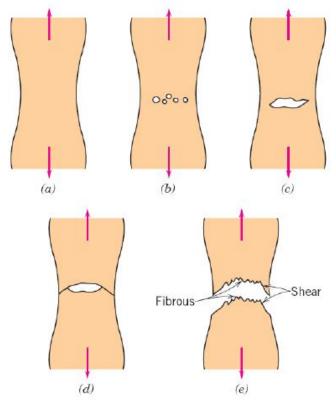
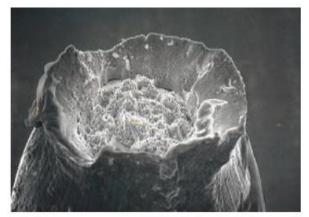


Figure 8.2 Stages in the cup-and-cone fracture. (a) Initial necking. (b) Small cavity formation. (c) Coalescence of cavities to form a crack. (d) Crack propagation. (e) Final shear fracture at a 45° angle relative to the tensile direction. (From K. M. Ralls, T. H. Courtney, and J. Wulff, Introduction to Materials Science and Engineering, p. 468. Copyright © 1976 by John Wiley & Sons, New York. Reprinted by permission of John Wiley & Sons, Inc.)



Fracture surface of tire cord wire loaded in tension. Courtesy of F. Roehrig, CC Technologies, Dublin, OH. Used with permission.

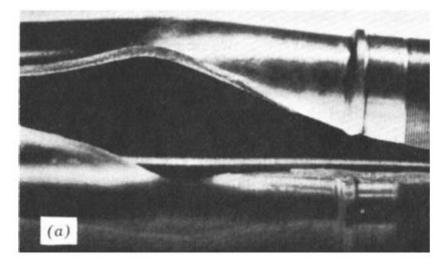




Examples of Ductile and Brittle Fracture of Pipes

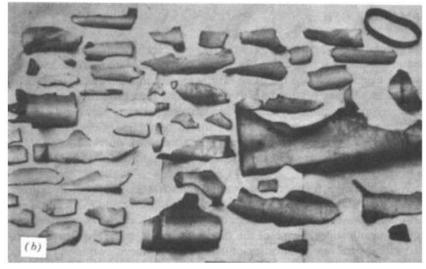
Ductile fracture:

- -- one piece
- -- large deformation



Brittle fracture:

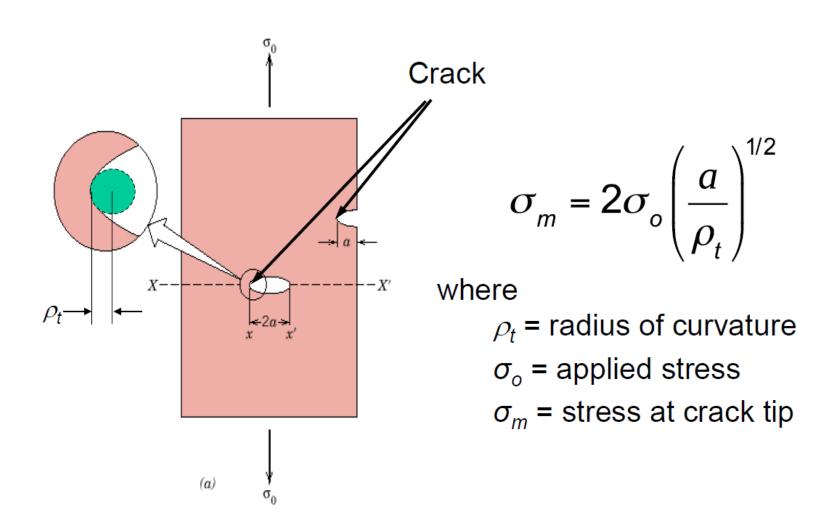
- -- many pieces
- -- small deformations







Fracture Mechanics



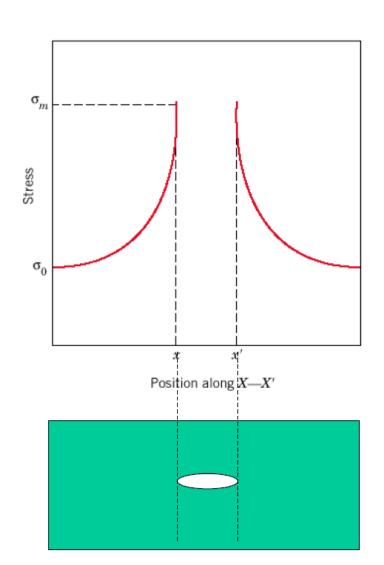


Fracture Mechanics

Stress Concentration at Crack Tip

 K_t = stress concentration factor

$$K_t = \frac{\sigma_m}{\sigma_o}$$

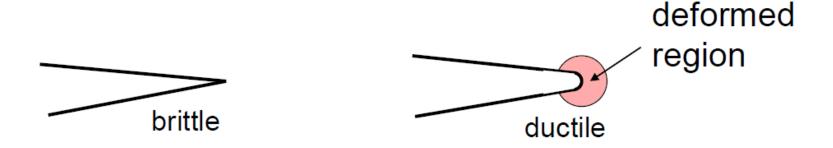






Crack Propagation

- Stress concentration higher for sharp cracks propagate at lower stresses than cracks with blunt tips
- For ductile materials—plastic deformation at crack tip when stress reaches yield strength—tip blunted lowers stress conc.

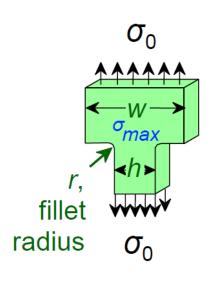




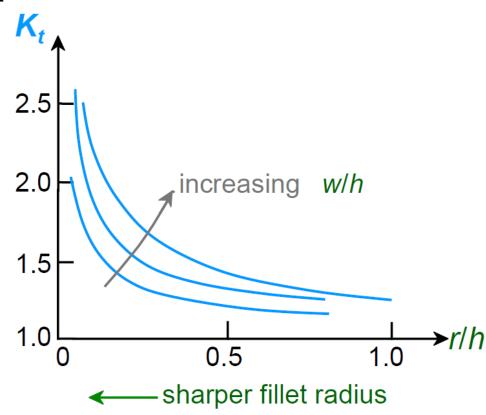


Crack Propagation

Avoid sharp corners!



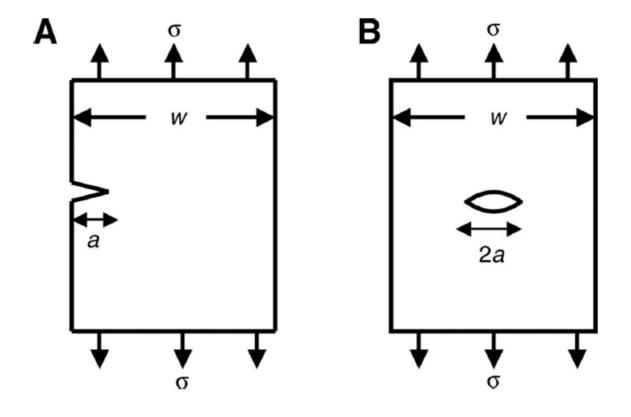
Adapted from Fig. 8.2W(c), Callister 6e. (Fig. 8.2W(c) is from G.H. Neugebauer, Prod. Eng. (NY), Vol. 14, pp. 82-87 1943.)







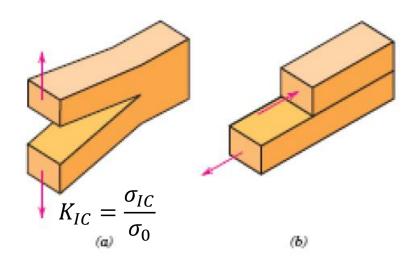
Fracture toughness





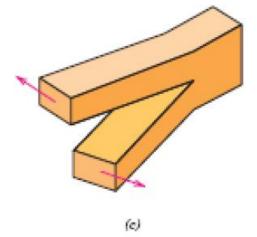


Fracture modes



Tensile mode

Sliding mode

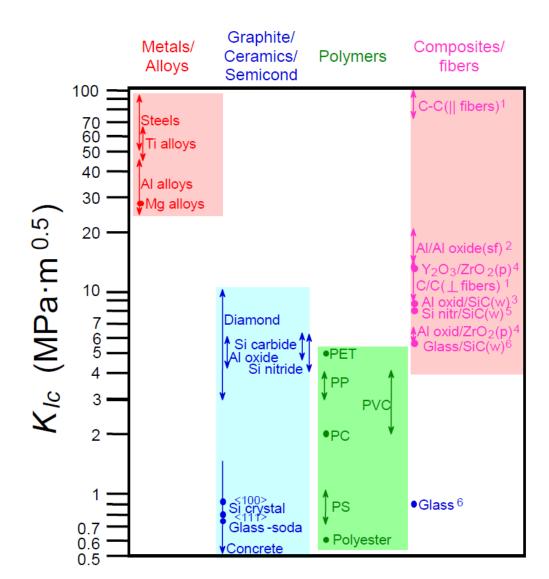


Tearing mode





Fracture Toughness Ranges







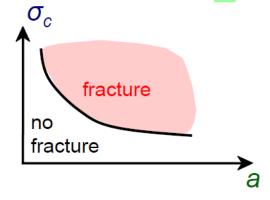
Design Against Fracture

Crack growth condition:

$$K_{lc} < Y_{\sigma_C} \sqrt{\pi a}$$

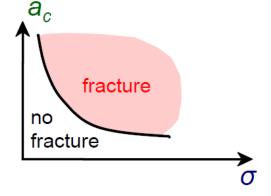
--Scenario 1: K_{lc} and flaw size *a* specified - dictates max. design (critical) stress.

$$\sigma_c = \frac{K_{Ic}}{Y \sqrt{\pi a}}$$



--Scenario 2: *K*_{lc} and stress level specified - dictates max. allowable flaw size.

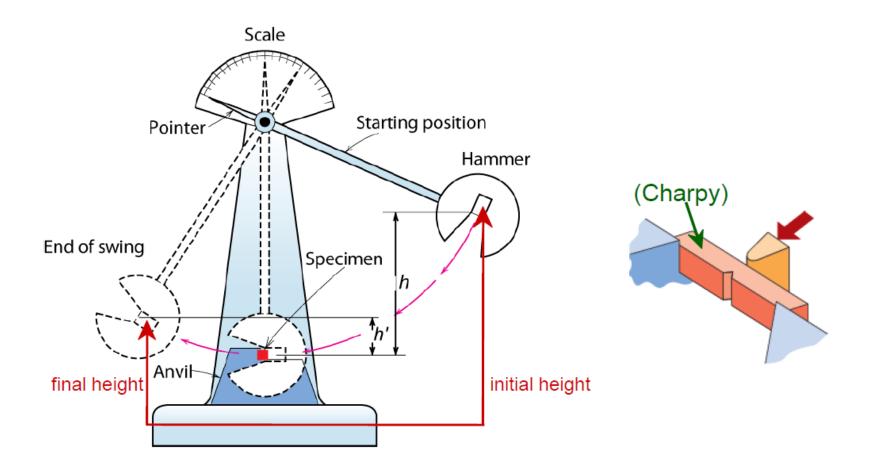
$$a_c = \frac{1}{\pi} \left(\frac{K_{Ic}}{Y \sigma} \right)^2$$







Impact fracturing test





Ductile-to-Brittle Transition

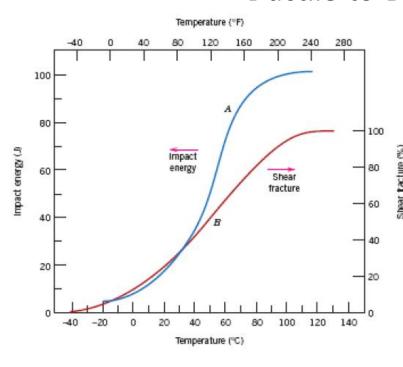
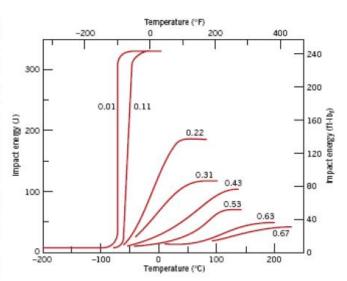


Figure 8.16 Influence of carbon content on the Charpy V-notch energy-versustemperature behavior for steel. (Reprinted with permission from ASM International, Metals Park, OH 44073-9989, USA; J. A. Reinbolt and W. J. Harris, Jr., "Effect of Alloying Elements on Notch Toughness of Pearlitic Steels," Transactions of ASM, Vol. 43, 1951.)



Brittle Fracture

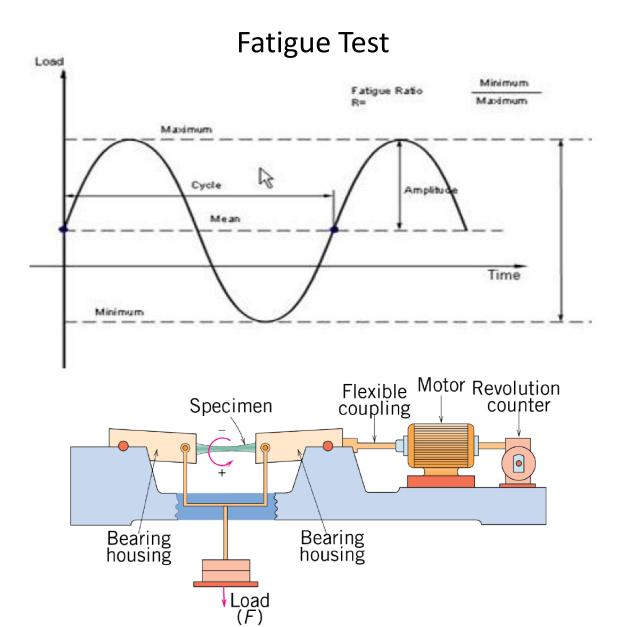


Ductile Fracture













Fatigue Failure

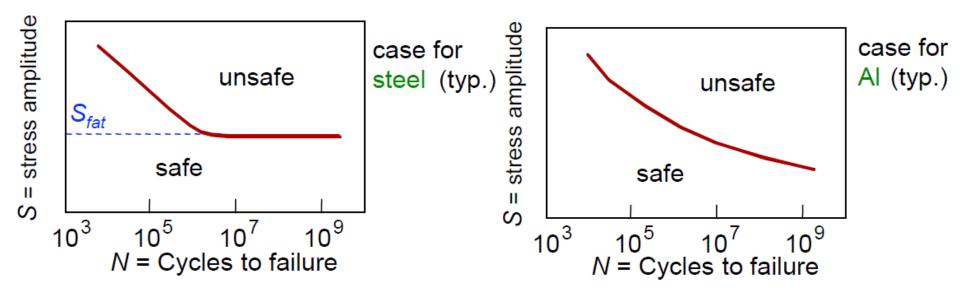


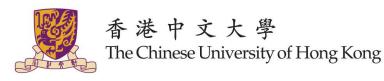






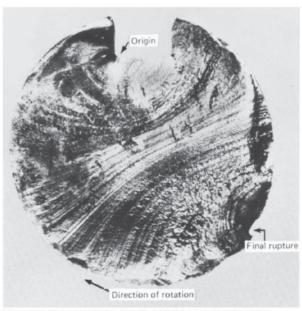
Types of Fatigue Behavior



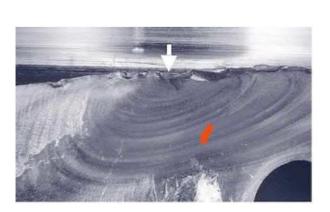


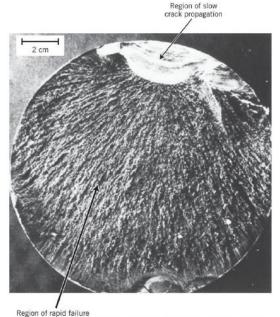


Types of Fatigue Behavior



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Reproduced by permission from *Metals Handbook: Fractography and Atlas of Fractographs*, Vol. 9, 8th edition, H. E. Boyer (Editor), American Society for Metals, 1974.

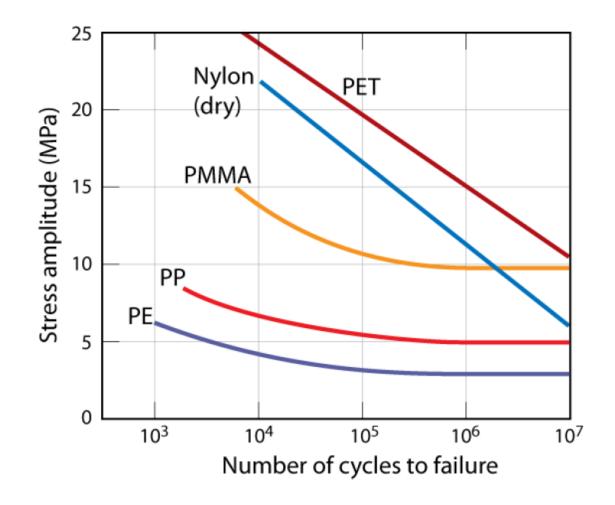




Fatigue Behavior of Polymers

- Fatigue limit:
 - PMMA, PP, PE

- No fatigue limit:
 - PET, Nylon (dry)

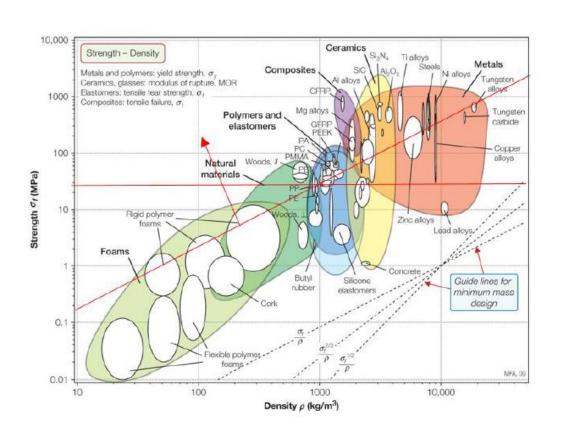


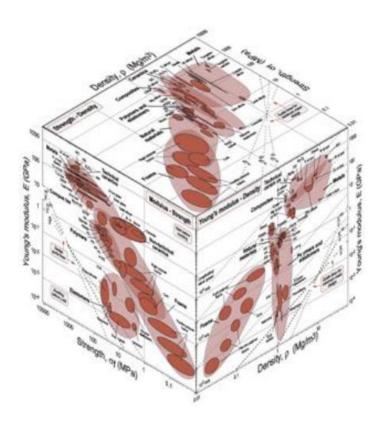


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